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Low temperature anelasticity in Ti6Al4V alloy and Ti6Al4V-SiC_f composite

S. Amadori^{a,*}, E. Bonetti^a, L. Pasquini^a, P. Deodati^b, R. Donnini^b, R. Montanari^b, C. Testani^c

^a Department of Physics, University of Bologna and CNISM, v.le Berti Pichat 6/2, 40127 Bologna, Italy

^b Department of Mechanical Engineering, University of Rome "Tor Vergata", Via del Politecnico 1, 00133 Rome, Italy

^c Centro Sviluppo Materiali (C.S.M.), Via di Castel Romano 100, 00128 Rome, Italy

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ABSTRACT

Internal friction and dynamic Young modulus measurements have been performed in the temperature range 80–350 K on the Ti6Al4V alloy and a Ti6Al4V–SiC_f composite reinforced by unidirectional SiC fibers (SCS-6). A vibrating reed apparatus operating at low strain amplitude (<10⁻⁵) in the frequency range 10²–10⁴ Hz was employed. Two anelastic relaxation peaks: P₁, P₂ have been observed in both materials. The peak temperatures at 1 kHz are respectively 120 K and 250 K. The fiber reinforced composite in the same temperature range shows a higher background damping and enhanced relaxation strength for the P₁ peak. The activation energy and frequency factors evaluated from the peak temperature shift with frequency are respectively $H_1 = 0.21 \pm 0.02$ eV, $H_2 = 0.50 \pm 0.03$ eV and $\tau_{01}^{-1} = 10^{15}$ s⁻¹. $\tau_{02}^{-1} = 10^{12}$ s⁻¹. Both peaks position and relaxation strength depend on hydrogen content and the relative amount of defects in α and β phases.

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1. Introduction

Titanium fiber reinforced composites are promising materials for aircraft industry [1,2]. Gaseous impurities such as hydrogen nitrogen and oxygen absorbed during manufacturing processes and in operating conditions can strongly modify their mechanical behaviour. In these composites further structural instability phenomena may occur at the fiber–matrix interfaces determined by diffusion processes and thermally driven chemical reactions [3–5].

The Ti6Al4V alloy, consists of an $(\alpha + \beta)$ two phase structure. The solubility of hydrogen in the bcc β phase is much higher than in the hcp α phase, the activation energy reported for hydrogen diffusion in the β phase is much lower than in the α phase [6]. Anelasticity measurements through mechanical spectroscopy have been extensively employed to understand details of point defects dynamics in Ti alloys [7] and to investigate transient or permanent damping effects depending on the evolution of thermal stresses at the fiber–matrix interfaces in composites [8]. This paper companion of another one [9] aims at a comparative investigation of the anelastic behaviour at low temperatures of the Ti6Al4V–SiC_f composite and the corresponding monolithic matrix alloy, to understand the role of the microstructure, defects content on anelasticity phenomena depending on hydrogen.

2. Experimental

The Ti6Al4V–SiC_f composite was produced at Centro Sviluppo Materiali (CSM) by HIP (Hot Isostatic Pressing) in a multistep process, starting from layers of unidirectional SCS-6 fibers in alternate sequence with Ti6Al4V sheets. Details of the processing steps are reported elsewhere [9]. The microstructure of the composite matrix mainly consisted of α phase with about 4% β phase. The monolithic alloy displays a similar duplex ($\alpha + \beta$) microstructure.

The inverse mechanical quality factor (internal friction) and the resonance frequency, proportional to the square root of the dynamic young modulus at constant density and shape, have been measured by a completely automated vibrating reed analyzer VRA-1604 (CANTIL Srl). The apparatus employs electrostatic excitation and frequency modulation detection of the flexural vibration modes of cantilevered mounted reed samples with typical dimensions: 6 cm length, 0.5 cm width, 0.08 cm thickness. All measurements of the internal friction (Q^{-1}) and of the resonance frequency (f), were performed in the 0.6-3.5 kHz frequency range at a strain amplitude $\varepsilon < 10^{-5}$ and a pressure in the vacuum chamber 10^{-4} Pa to 10^{-3} Pa. Before each measurement run, in the 80-350 K range at a constant heating rate of 1 K/min, all samples where thermally equilibrated at the lower temperature for 30 min. All ageing treatments at temperatures in the 600-900 K range have been performed in a vacuum of 10⁻⁴ Pa. To gain a clearer understanding of some specific features of the internal friction spectra, some measurements were performed also on a commercial β rich α + β titanium alloy SP-700 (Ti-4.5Al-3V-2Mo-2Fe).

^{*} Corresponding author. E-mail address: amadoristefano@libero.it (S. Amadori).

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Fig. 1. Internal friction and normalized squared resonance frequency of the monolithic alloy and the composite as prepared. f_0 = resonance frequency at the lowest temperature at the start of the heating run: ~600 Hz for Ti6Al4V and ~900 Hz for the Ti6Al4V + SiC composite.

3. Results

Fig. 1 shows the internal friction and the squared resonance frequency (normalized to the value at the lower temperature at the start of measurement: $(f/f_0)^2$), versus temperature for the monolithic alloy and the as prepared fiber reinforced composite. A prominent peak thereafter indicated P₁, is observed in both materials at approximately 120 K (1 kHz), the relaxation strength is significantly higher in the composite in comparison with the monolithic alloy. A second very broad peak P₂ centered at 250 K in the alloy and 270 K in the composite spectra has been observed. The background damping in the whole temperature range investigated appears higher in the composite.

The relaxational origin of both peaks was confirmed by the shift of the peaks temperature with frequency. The values obtained for the activation energy and frequency factors are reported in Table 1. After thermal ageing up to 900 K in a 10^{-4} Pa vacuum, a significant reduction of the P₁ peak relaxation strength both in the monolithic alloy and the composite, up to 50%, is experienced, as can be seen in Fig. 2.

In the composite the background damping with respect to the as received samples (Fig. 1) increases slightly with temperature, moreover a small reduction of the modulus (frequency) occurs with reference to the values experienced in as fabricated condition (Fig. 1). The P₁ peak with a relaxation strength similar to that of the composite and significantly higher than that of the monolithic matrix alloy is observed also on the SP-700 alloy (Fig. 3).

4. Discussion

On the basis of the internal friction spectra presented above, the P₁ peak can be attributed to hydrogen. The activation energy obtained from the peak shift (Table 1) is slightly lower than that reported for hydrogen diffusion in the β phase of vanadium free titanium [6], but is slightly higher than those obtained by NMR in titanium with vanadium additions [10]. A value comparable with our results, within the uncertainty of measurements, was obtained by internal friction measurements on the Ti6Al4V alloy [11]. In the

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	Anelastic ı	relaxation	peaks	parameters
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Peak	P ₁	P ₂
Temperature at 1 kHz frequency (K) Activation energy (eV) Inverse frequency factors (s ⁻¹)	$\begin{array}{c} 120 \\ 0.21 \pm 0.02 \\ 10^{15} \end{array}$	$\begin{array}{c} 250{-}270\\ 0.50{\pm}0.03\\ 10^{12} \end{array}$



Fig. 2. Internal friction and resonance frequency of the same samples as in Fig. 1 after ageing at 900 K in vacuum. f_0 resonance frequency of Fig. 1.

SP-700 alloy (Fig. 3), a well resolved peak having roughly the same activation energy of that in the Ti6Al4V alloy is present. Measurements performed at low frequency [12] yield similar activation energy. The prominent P_1 peak can be therefore attributed to a Snoek type relaxation caused by interstitials hydrogen atoms in the β phase.

A significant difference in the P₁ relaxation strength is yet observed (Fig. 3) between the matrix alloy and the composite. Moreover the peak relaxation strength in the SP-700, mainly β phase alloy, is similar to that of the composite (Fig. 3). The overall hydrogen content, measured by means of chemical analysis, was respectively: 116 ppm in the monolithic Ti6Al4V alloy, 187 ppm in the composite and 97 ppm in the SP-700 alloy. These values are within the low concentration solubility limit where a linear dependence between the relaxation strength and the amount of interstitial hydrogen responsible for Snoek relaxation in the β phase is observed [12]. It is well known that in two phase $(\alpha + \beta)$ titanium alloys hydrogen mainly accumulates in β phase; the relative amount of hydrogen in α/β interfaces and in α phase is very small [13,14] so the concentrations in the β phase are close to the overall contents reported above. Therefore, the larger relaxation strength in SP700, with lower hydrogen content, than in Ti6Al4V appears surprising. Moreover, the ratio between relaxation strengths of Ti6Al4V and composite does not correspond to that between their hydrogen contents. The apparent inconsistency of hydrogen contents and relaxation strengths can be explained by assuming that not all the



Fig. 3. Comparison of the internal friction spectra in the temperature range of the P_1 peak for the composite, the alloy and SP-700. Background damping has been subtracted. The P_1 peak is broader with respect to that resulting by assuming a Debye relaxation.

hydrogen atoms present in the β phase participate to relaxation processes because a part of them are trapped by lattice defects. This hypothesis is supported by the values of dislocation densities determined by XRD which are 5.5×10^{10} cm⁻², 6.1×10^9 cm⁻² [9] and 2.5×10^9 cm⁻² respectively for Ti6Al4V, composite and SP700.

The relaxation strength reduction after ageing, which is of the same order (50%) in the Ti6Al4V alloy and in the composite for similar ageing treatments (Figs. 1 and 2), is consistent with a similar mechanism of hydrogen occupancy reduction of the interstitial sites in the β phase, responsible for the P₁ Snoek type relaxation. No appreciable defect recovery was observed after long-term treatments at 873 K, temperature close to that of ageing [15,16].

Present measurements show that the P_1 peak is much broader than a Debye peak (Fig. 3). The broadening effect is larger in the composite. A possible effect of relaxation time distribution connected to microstrain and lattice distortion is under investigation.

The P₂ peak observed at around 250 K, has an activation energy, Table 1, corresponding within the measurement uncertainty, to that reported for hydrogen diffusion in the α phase [6]. A peak with similar activation energy was observed in Ti6Al4V [11] and ascribed to hydrogen in α phase. The mechanism proposed is a stress-induced reorientation of interstitial-substitutional pairs in the α phase according to the model proposed by Gupta and Weining [17] for hcp metals.

5. Summary

Comparative measurements at low temperatures on the Ti6Al4V–SiC_f composite and the corresponding monolithic alloy showed two anelastic relaxation peaks, P_1 , P_2 :

 P_1 results from a Snoek type relaxation due to interstitial hydrogen in the minority bcc β phase. Relaxation strengths are affected by the higher dislocation density in alloy than in composite P_2 has been ascribed to stress-induced reorientation of interstitialsubstitutional pairs in the hcp α phase.

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References

- S. Nourbakhsh, H. Margolin, in: R.B. Bhagat, et al. (Eds.), Fabrication of high temperature fiber reinforced intermetallic matrix composites, Metal & Ceramic Matrix Composites: Processing, Modeling and Mechanical Behavior, TMS, 1990, pp. 75.
- [2] S.C. Jha, in: S.K. Das, C.P. Ballard, F. Maribar (Eds.), High Performance Composites for the 1990s, TMS, Warrendale, PA, 1991, p. 459.
- [3] M.E. Tata, R. Montanari, C. Testani, G. Valdrè, La Metallurgia Italiana 7–8 (2005) 43.
- [4] R. Donnini, S. Kaciulis, A. Mezzi, R. Montanari, M.E. Tata, C. Testani, La Metallurgia Italiana 10 (2007) 15.
- [5] A. Mezzi, R. Donnini, S. Kaciulis, R. Montanari, C. Testani, Surf. Interface Anal. 40 (2008) 277.
- [6] R.J. Wasilewsky, G.L. Kehl, Metallurgia 50 (1954) 225.
- [7] M.S. Blanter, I.S. Golovin, H. Neuhauser, H.R. Sinning, Internal Friction in Metallic Materials—A Handbook, Springer, Berlin, 2007, p. 261.
- [8] R. Schaller, Mater. Sci. Eng. A 442 (2006) 423.
 [9] P. Deodati, R. Donnini, R. Montanari, C. Testani, High temperature damping Behaviour of Ti6Al4V–SiC_f Composite, Proceedings of the 15th International conference of Internal Friction and Mechanical Spectroscopy.
- [10] C. Korn, D. Teitel, Phys. Status Solidi (a) 44 (1977) 755.
- [11] Du Jia Ju, J. Physique, C5 42 (1981) 775.
- [12] X.S. Guan, H. Numakura, M. Koiwa, K. Hasegawa, C. Ouchi, Mater. Sci. Eng. A 272 (1999) 230.
- [13] G.F. Pittinato, W.D. Hanna, Met. Trans. 3 (1972) 2905.
- [14] H. Saitoh, T. Ohnishi, J. Mater. Sci. Lett. 14 (1995) 417.
- [15] R. Donnini, S. Kaciulis, A. Mezzi, R. Montanari, C. Testani, Mater. Sci. Forum 604–605 (2009) 331–340.
- [16] P. Deodati, R. Donnini, R. Montanari, C. Testani, T. Valente, Mater. Sci. Forum 604–605 (2009) 341–350.
- [17] D. Gupta, S. Weining, Acta Metall. 10 (1962) 292.